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NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

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VIBRATION SURVEYS OF THE P-40 RUDDER

AND FIN-RUDDER ASSEMBLY

By Theodore Theodorsen and Arthur A. Regier

Langley Memorial Aeronautical Laboratory Langley Field, Va.



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MEMORANDUM REPORT

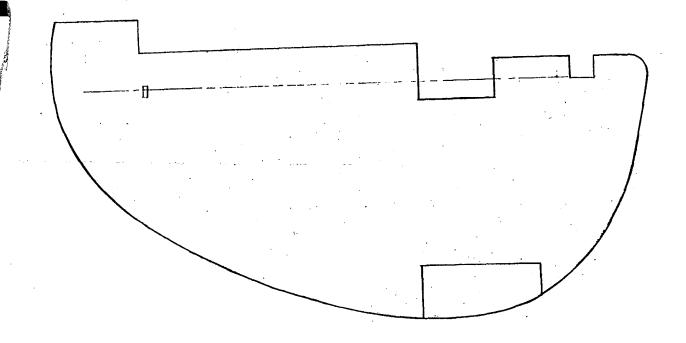
for the

Army Air Forces, Materiel Command
VIBRATION SURVEYS OF THE P-40 RUDDER
AND FIN-RUDDER ASSEMBLY

By Theodore Theodorsen and Arthur A. Regier

The following surveys on the vibration response of the P-40 rudder and fin-rudder assembly were conducted during the fall of 1942 for the purpose of obtaining information on the flutter characteristics of high-speed airplanes of known performance. This work is a continuation of similar study on the P-47 airplane. The tests on the rudder alone were made with the rudder suspended in a floating condition; the tests on the fin-rudder unit were made with the fin rigidly fastened to a concrete base. In the figures the plus and minus signs are used to signify opposite phases with the size of each sign proportional to the amplitude. In several figures the nodal lines are drawn.

Langley Memorial Aeronautical Laboratory, National Advisory Committee for Aeronautics, Langley Field, Va., April 28, 1943.



I. P-40-F FABRIC RUDDER

Part No. 87-14-60, 54-603

Natural frequences (suspended in rudder with tab loose): 31.5, 47, 69, 85, 110, 125

Frequency, Remarks cycles per second

31.5 47	Lowest bending 2 nodes. Bending 2 nodes approximately perpendicular to
69	nodes of lowest bending. Torsion
•	dicular to hinge line.
85	Bending 4 nodes.
110	Local bending and torsion.
125	Local vibration. NATIONAL ADVISORY COMMITTEE FOR AERONAUTIC:

II. FIN AND RUDDER ASSEMBLY FOR CURTISS P-40-F AIRPLANE

	Metal fin data	Fabric rudder data
Part No. 87-12-50, MF 1208	8'	7-14-60, 54-603
Weight Maximum height Maximum chord Maximum thickness	46 inches 37 inches	26 pounds 74 inches 35 inches

Vibration frequencies as mounted on concrete:

Fin alone....bending 85 cycles per second

Frequencies of assembly as mounted on concrete:

Frequency, cycles per second

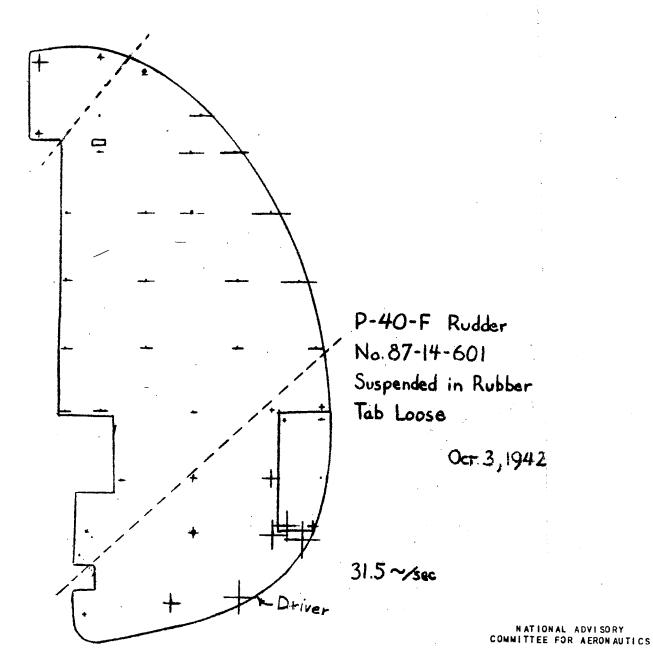
125

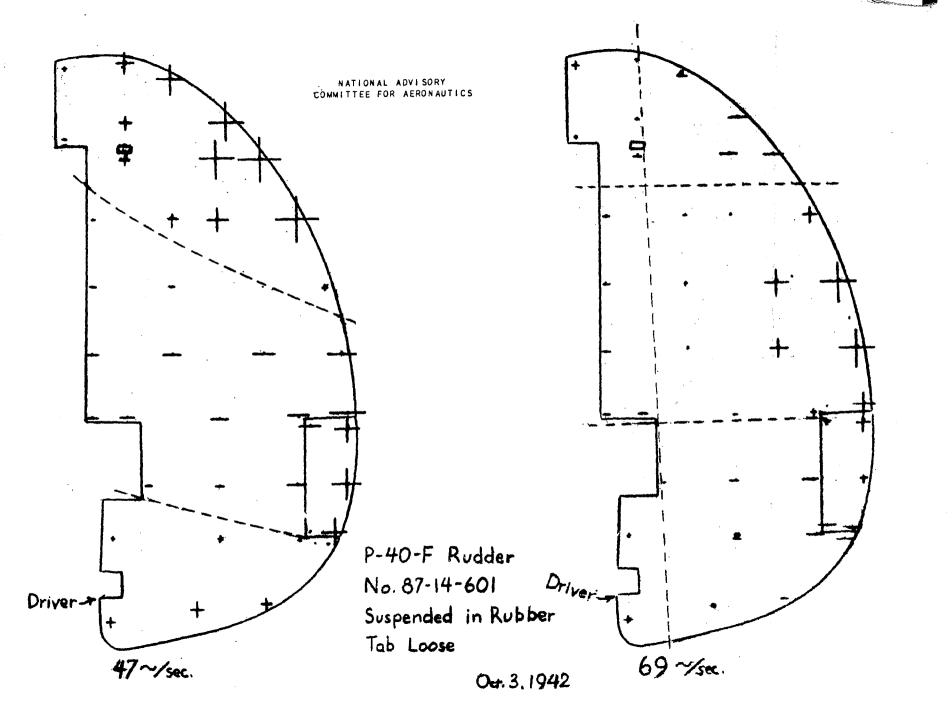
21.5, 37.5, 56, 62, 82, 91, 101, 107, 125

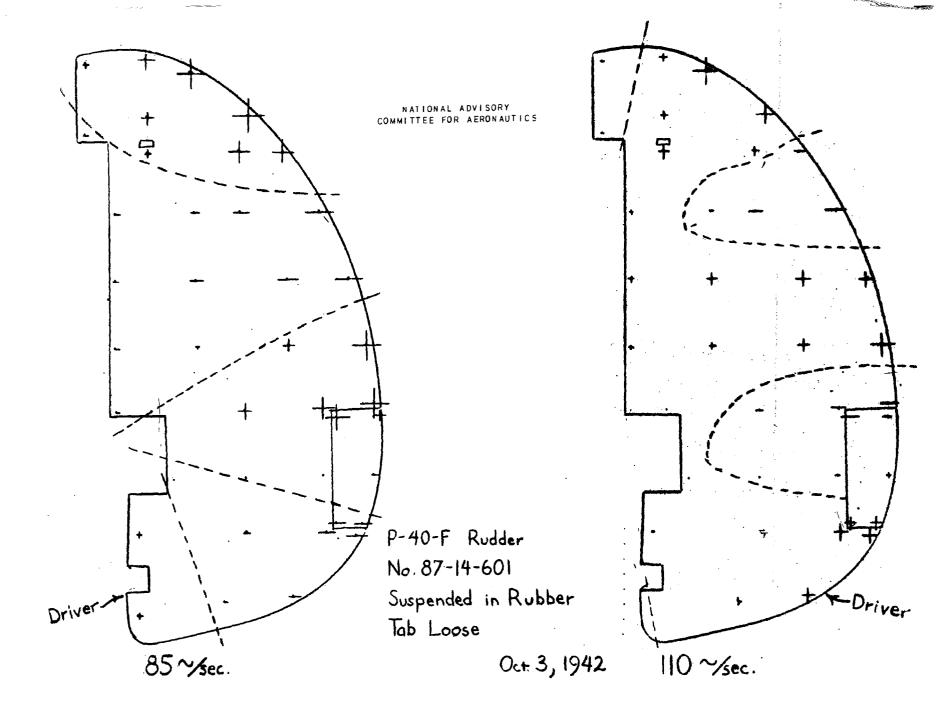
Remarks

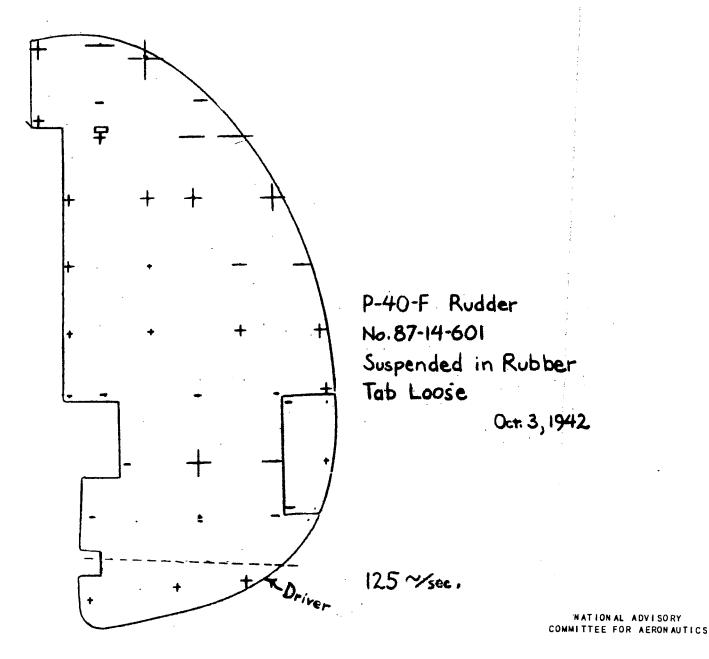
8 5	First bending of fin alone.
21.5	First bending of combination. No nodes except at support.
37.5	Bending with one node in rudder and node at top counterweight.
56	Torsion with one node near hinge line and two nodes perpendicular to hinge line.
62	Same as above except that fin has reversed
•	phase with respect to the rudder.
82	Bending with three nodes.
91	First bending of fin with three nodes in rudder.
101	Fin bending with local rudder responses.
107	Local vibration.
± • •	TO OUT AT STORY

Local vibration.

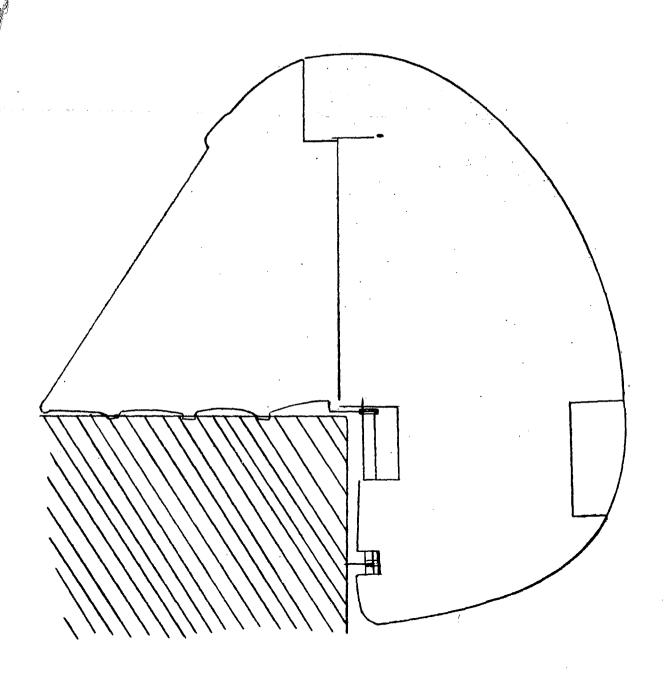




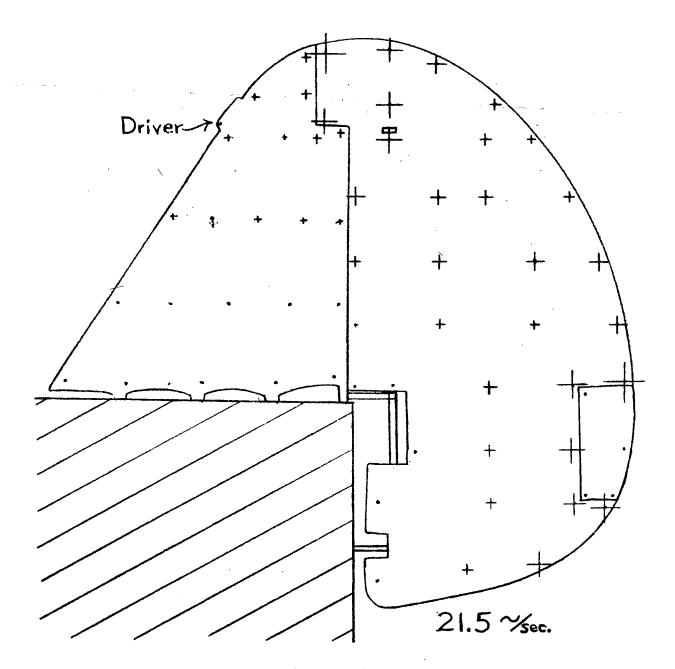




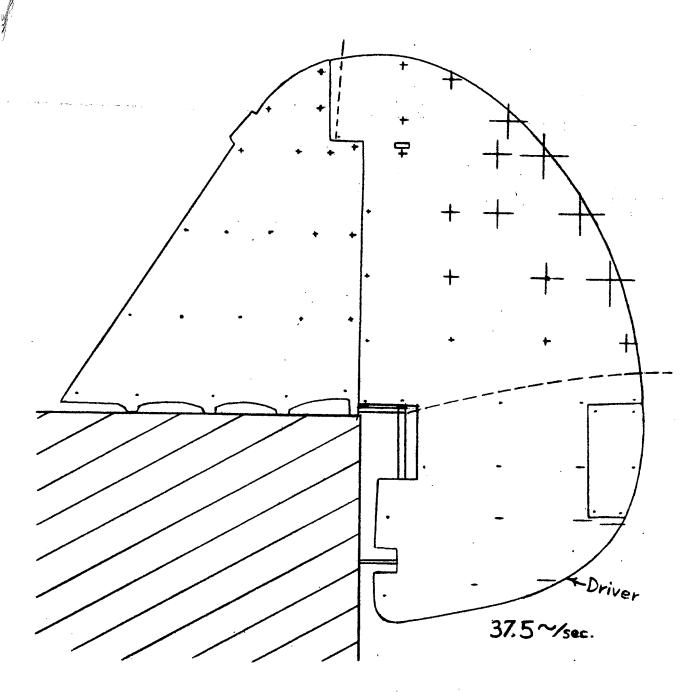
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Fin and rudder assembly for Curtiss P-40-F airplane.



P-40 Rudder Attached to Fin on Concrete Block, Tab Loose Oct. 6, 1942

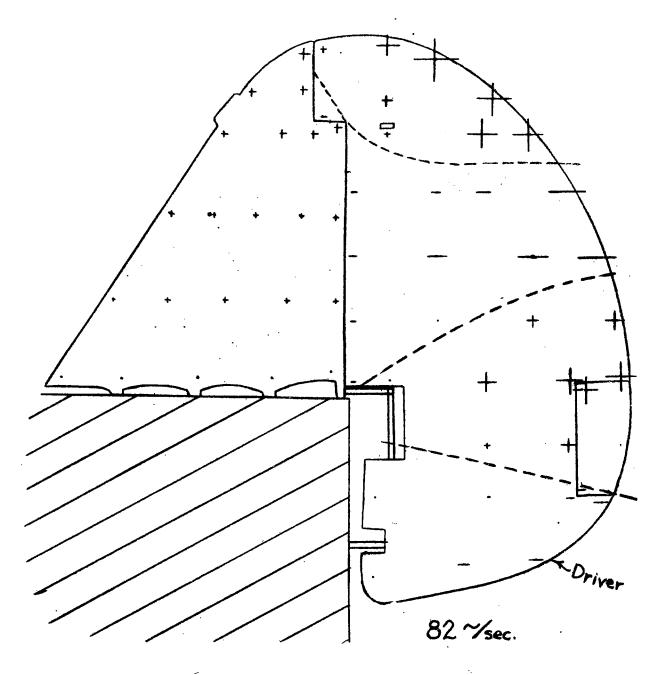


P-40 Rudder Attached to Fin on Concrete Block, Tab Loose Oct. 5, 1942

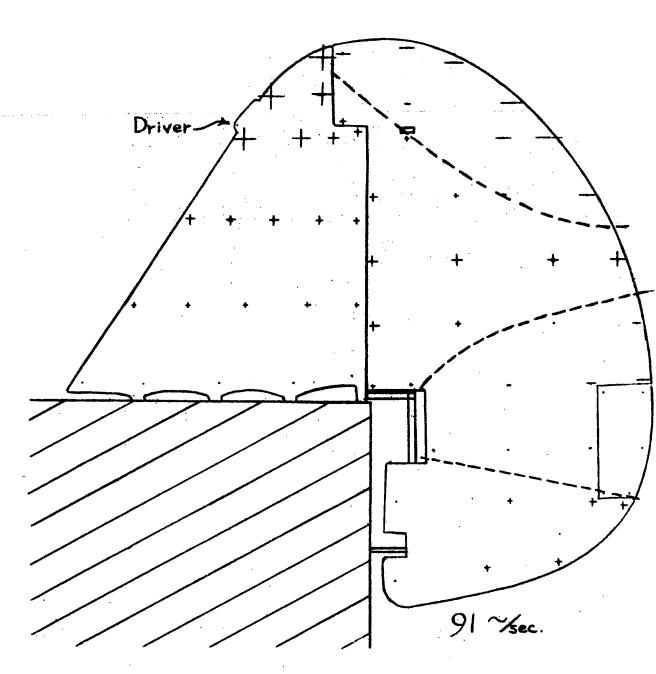
P-40 Rudder Attached to Fin on Concrete Block, Tab Loose Oct. 5, 1942

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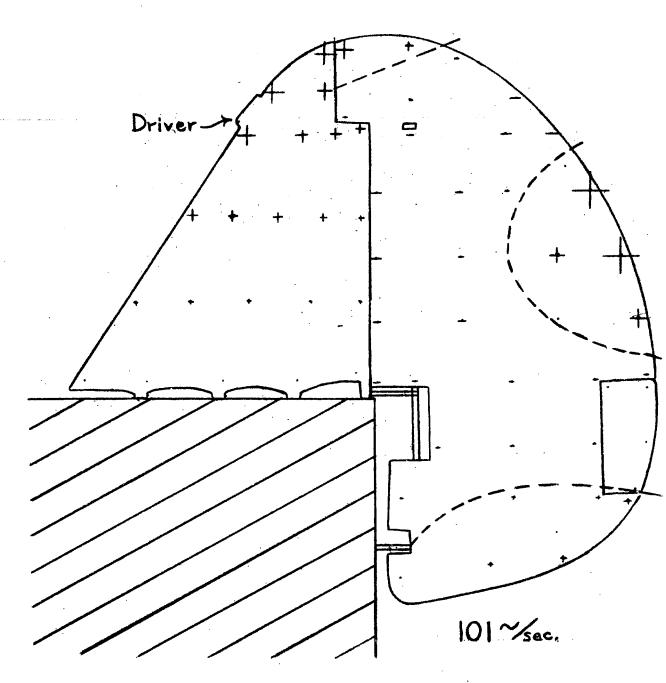


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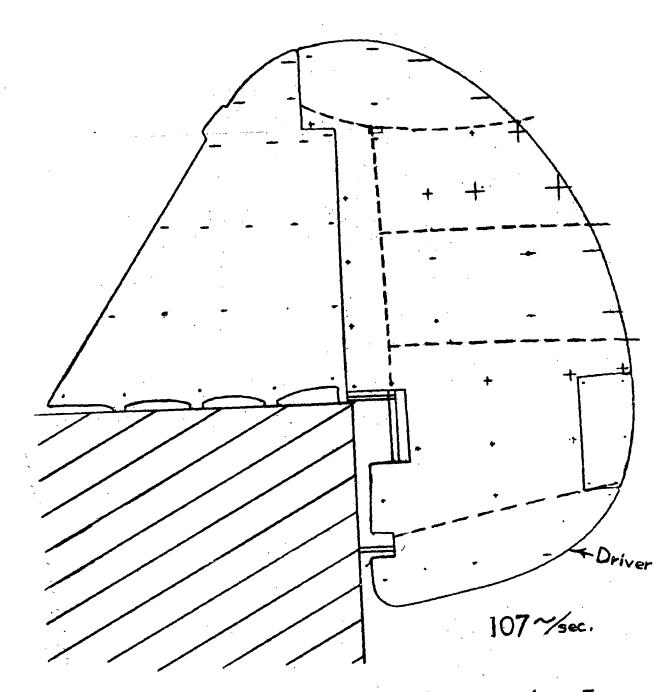
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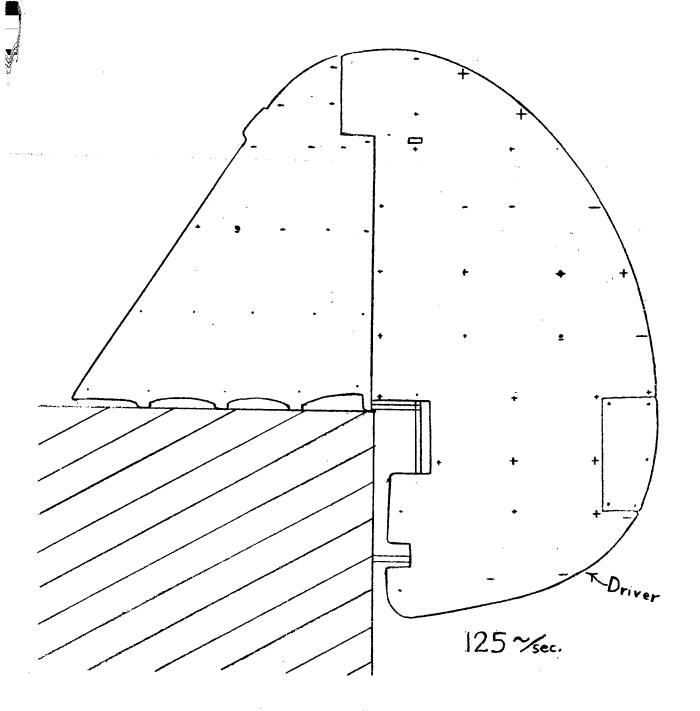
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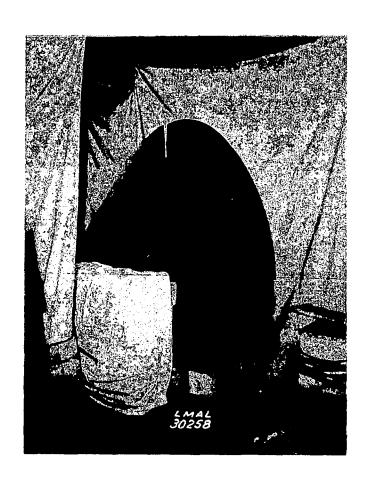


P-40 Rudder Attached to Fin on Concrete Block, Tab Loose Oct. 5, 1942

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P-40 Rudder Attached to Fin on Concrete Block, Tab Loose Oct. 5, 1942



P-40 rudder and fin-rudder assembly.

